M. KHOROLSKYI¹, S. BIGUN²

RUBBER AS AN ADVANCED STRUCTURAL MATERIAL FOR THERMAL CONDITIONING SYSTEM INTERFACES

¹Oles Honchar Dnipro National University 72 Haharina Ave., Dnipro 49010, Ukraine;e-mail: khomis@outlook.com ²Yuzhnoye State Design Office 3 Kryvorizka St., Dnipro 49008, Ukraine; e-mail: bigun3008@gmail.com

In integrated launch vehicles, one of the systems responsible for successful launch preparation and support is a ground thermal conditioning system supplying low-pressure thermostatic air to the "dry" compartments and head blocks of a launch vehicle. To connect the thermal conditioning system to the launch vehicle, a special interface is used. The proper functioning of the interface is critical to the reliability of the ground equipment of the system, the launch vehicle, and the space complex as a whole. This article describes key requirements to the interfaces of the thermal conditioning system and the drawbacks of their existing designs. The article proposes a new concept of interface design, according to which the pipeline of the ground thermal conditioning system is connected to the inlet tube of the launch vehicle via a corrugated rubber hose composed of three basic parts. The hose is attached to the inlet tube of the launch vehicle with the help of a metal lock/unlock device. The proposed solution provides good air tightness, ease of operation, easy multiple connections to the launch vehicle at different angles, and an automatic disconnection at launch or a manual disconnection in the case of a cancelled launch. Using rubber, which is a high-elasticity structural material, in the manufacturing of hoses makes it possible to minimise the effort required to disconnect the interface from the launch vehicle. In a high elasticity state, rubber can absorb and dissipate mechanical energy over a wide range of temperatures, which precludes the vibration caused by the engine operation from being transmitted to the ground thermal conditioning system. The article presents the key properties of rubber used as a structural material and its features to be considered in the design of similar devices. In contrast to metal, which shows two types of deformation (elastic and plastic), rubber can exhibit three types (elastic, superelastic, and plastic). During the design of interfaces, two types of deformation were taken into account: elastic and superelastic. Experimental tests of the interface presented in the article showed its full compliance with the requirements specification.

Keywords: vehicle inlet tube, corrugated rubber horse, lock/unlock device, superelastic deformation, air tightness.

1. Bigun S. A., Yelansky Yu. A., Khorolsky M. S. Types and design features of interfaces of launch vehicle head block and compartment thermal conditioning systems. Space Technology. Missile Armaments. 2013. Iss. 1. Pp. 65-68. (in Russian).

 Bigun S. A., Khorolsky M. S. Topical issues of creation of space rocket thermostatic system mating points. Space Technology. Missile Armaments. 2018. Iss. 2. Pp. 132-138. (in Russian). https://doi.org/10.33136/stma2018.02.132

3. Brevet d'invention français No. 2658479 (2), 1991, IPC B64G 1/40; B64G 1/64; B64 G 5/00.

4. Brevet d'invention français No. 2685903 (1), 1993, IPC B64G 5/00; F41F3/055; F02K9/44.

5. Russian Federation Patent No. 2473003-S1, 2011, IPC7 F16L 37/20. (in Russian).

6. Yurtsev L. N., Bukhin B. L. Rubber as a Structural Material. Rubberman's Comprehensive Handbook. In two parts. Part 1. Rubbers and Ingredients / S. V. Reznichenko, Yu. L. Morozov (Eds.). Moscow: MAI Tekhinform, 2012. 744 pp. (in Russian).

7. USSR State Standard GOST 263-75. Rubber. Shore A Hardness Measurement Method (with changes No. 1, 2, 3, and 4). oscow: Standards, 1989. 10 pp. (in Russian).

8. Koshelev F. F., Kornev A. E., Bukanov A. M. General Technology of Rubber. Moscow: Khimiya, 1978. 528 pp. (in Russian).

9. Skokov A. I., Kaplun S. V., Bogutskaya E. A., Khorolsky M. S., Bigun S. A. Technological aspects of the development of connection hoses for thermal conditioning systems of launch vehicles. Space Technology. Missile Armaments. 2015. Iss. 1. Pp. 42-45. (in Russian).

10. Bigun S. A., Yevchik V. S., Khololsky M. S. On selection of materials for creation of modern LV thermostating system mating hoses. Space Technology. Missile Armaments. 2018. Iss. 1. Pp. 72-84. (in Russian). https://doi.org/10.33136/stma2018.01.072

11. Ukrainian Patent No. 120445, 2019, B64G 5/00, B64G 1/40, F16L 37/08, F41F 3/055, F16L 33/00; S. O. Bigun, I. P. Babych, O. V. Nesterov, M. S. Khorolskyi, O. I. Skokov. Application No. a 2017 11016; filed on November 10, 2017; published on December 10, 2019, Bul. No. 23, 4 pp. (in Ukrainian).

12. Ukrainian Patent No. 120469, 2019, B64G 5/00, B64G 1/40, F25B 29/00, F16L 33/00, F16L 37/12, F16L 25/00; S. O. Bigun, M. S. Khorolskyi. Application No. a 2018 02244; filed on March 5, 2018; published on December 10, 2018, Bul. No. 23, 4 pp. (in Ukrainian).

13. Khololskyi M. S., Bigun S. O. On a concept of the development of interfaces of thermal conditioning systems for integrated launch vehicles. In: Aerospace Hardware System Design and Performance Analysis. V. XXVII. S. O. Davydov (Ed.). Dnipro: Lira, 2019. Pp. 162-168. (in Ukrainian).

14. Belyaev N. M. Strength of Materials. Moscow: Nauka, 1965. 856 pp. (in Russian).

15. Bigun S. A., Khorolsky, M. S., Scokov A. I., Kaplun S. V. Experimental investigations into results of testing Cyclone-4 ILV thermostating system mating points. Space Technology. Missile Armaments.,2016. Iss. 2. Pp. 43-51. (in Russian).

Method for the manufacturing of connection hoses for the pipeline of a thermal conditioning system. Patent No. 143346, Ukraine: IPC B64G 5/00, B64G 1/40, F16L 37/08, F16L 33/00, F41F 3/055; S. O. Bihun, I. P. Babych, O. V. Nesterov, M. S. Khorolskyi, O. I. Skokov, S. V. Kaplun. Application No. u 2020 00486; filed on January 28, 2020; published on July 27, 2020. Bul. No. 14. 4 pp. (in Ukrainian).

17. Khorolskyi M., Bigun S. Peculiarities of manufacturing of hoses for interfaces of thermal conditioning systems of integrated launch vehicles. Journal of Rocket-Space Technology. 2020. V. 28. No. 4. Pp. 62-69. (in Ukrainian).

Received on February 24, 2021, in final form on March 31, 2021